

# AVIATION

AND

## AIRCRAFT JOURNAL

DECEMBER 27, 1920

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# AVIATION AND AIRCRAFT JOURNAL

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### Marking Landing Fields

**O**NE of the crying needs of commercial aviation is a system of ground markings that will serve as aerial sign posts. One of the most logical places to display these markings is on landing fields and airports as they not only serve those looking for these points but also those that are merely passing overhead. To be of maximum utility such marks should conform to some uniform and simple system and should convey as much information as is practicable.

As the airport is essentially an international arena of transportation, Annex F of the International Air Convention contains provisions for a uniform system of ground marks that is admirably adapted to its purpose. The basis of this system is the International 1:2,500,000 scale map adopted by an international congress before the war. This map is divided into sheets comprising six degrees of longitude and four degrees of latitude. The ground marks consist of an open rectangle whose short sides shall be oriented north and south with the open side facing the opposite side of the east sheet. Within this rectangle is a dot indicating the relative position of the field on that half of the east sheet of the map. On the sides of the rectangle are figures indicating which way about the field is on.

After the airplane the Air Service started a system of marking fields in this country based on the state in which the field was located. The mark consisted of a letter and a small number. This system has never been employed to any great extent. It is simply an identification and does not give any further information than the name of the field is alone.

The international system on the other hand gives the navigator an indication of his position on the map without looking up the code to determine the name of the field and then finding it on the map. It may be argued that a navigator will know the meaning of the code and the position on the map. This is only true in territory that the navigator is familiar with and as aviation progresses the number of ground markings will increase greatly, rendering it more and more difficult to remember these locations by some arbitrary code.

The defect of the international system lies in the lack of unity of the field. If some code of letters or letters and figures were substituted for the numbers indicating the sheet, the uniformity would be proportionately increased. At large airports the name will undoubtedly be spelled out in magnetic letters but this is not practicable for every little landing field.

### Automobile Laws and Aerial Laws

**T**HE National Conference on Highway Traffic Regulations will meet in Washington on January 18 to reach final agreement on traffic regulations which can be presented with their endorsement to the security at large and particularly to the forty-two state legislatures concerning that month.

It is expected that the passage of these uniform laws will have the following effects:

First. To reduce deaths and injuries resulting from automobile accidents caused by ignorance or misinterpretation of traffic laws.

Second. To relieve the non-rapid courts which in many of the larger cities are now flooded with traffic violation cases.

Third. To save millions annually in property damage resulting from accidents due to violation or misinterpretation of traffic laws.

Fourth. To speed up traffic in the congested districts of cities by making it possible for the motorist to know what to do and what to expect from other motorists.

This represents an effort to overcome the confusion resulting from the variety of state motor laws which are now afflicted with. It is a lesson for aviation in that it shows the bad results of a lack of uniform law in a very similar field. If the word airplane be substituted for automobile, pilot for motorist, and airports for cities in the last case, the analogy would be as applicable as an argument for a federal aerial traffic law.

### The Need of Test Pilots

**I**t is a curious fact that many pilots develop an aversion for a certain type of machine without being able to state their reasons. When a statement is demanded they usually reply that the ship "doesn't feel right" and that some part "doesn't look right." It makes no difference to them if the machine has been tested in many ways they only are convinced that something is wrong somewhere.

On the other hand there are a few pilots who are able to fly a type for a time and then give a constructive criticism of the merits of the type. The majority of this small class do not seem to be particularly good pilots at first sight. They seem to be hesitantly fussy and not over anxious to fly. This is a wonderful experience, they appear to be fearful because they really know what shape a machine should be, and not over anxious to fly because they already have a great deal of flying time and more flying is just so much more work.

Many engineers are inclined to feel that comments from a pilot concerning their designs are somewhat of an annoyance. This is because they do not appreciate the value of a good test pilot. Aeronautical engineering can not depend absolutely on other wind tunnel or road load tests and hence must get a final and very accurate check from flight tests. These flight tests are carried out by the test pilot and his suggestions are very often of more constructive value than the data he collects. An experienced test pilot acquires a sort of intuition that tells him what will go and what will not. If his criticisms are given their proper weight in shaping a design the progress of aeronautical engineering would proceed much more rapidly.

# Air Service Requests \$60,000,000

The Chief of Air Service has submitted to Congress its estimated requirements for the fiscal year ending June 30, 1923. The personnel required consists of 1,514 officers and 16,000 enlisted men. The financial requirements are divided into eight groups shown below, the total of which amounts to \$60,000,000:

## ESTIMATE FOR FISCAL YEAR ENDING JUNE 30, 1923

### EXPENSES OF CIVILIAN EMPLOYEES

	Operation	Experimental Flight and Research	Total
Pay of civilians in the office of the Chief of Air Service	\$ 133,559.94	\$ —	\$ 133,559.94
Pay of civilians in the field	6,900,130.64	2,013,800.00	8,913,930.64
Traveling expenses of civilians on rolls in office of the Chief of Air Service	4,500.00	—	4,500.00
Traveling expenses of civilians on rolls in office of the Chief of Air Service	30,000.00	—	30,000.00
Pay of consulting engineers	4,500.00	—	4,500.00
Traveling expense of consulting engineers	500.00	—	500.00
<b>Totals</b>	<b>\$7,073,090.58</b>	<b>\$2,013,800.00</b>	<b>\$9,086,890.58</b>

### INSTRUCTION AND OPERATION OF AIR SERVICE TROOPS

	\$	\$	\$
Hydrogen, purchase of	300,000.00	—	300,000.00
Aircraft and their spare parts, maintenance and repair of	1,836,537.64	700,000.00	2,536,537.64
Engines and their spare parts, maintenance and repair of	596,730.00	—	596,730.00
Radars and their spare parts, maintenance and repair of	10,450.00	—	10,450.00
Engines and their spare parts, M & R	1,439,212.90	100,000.00	1,539,212.90
Instruments and accessories for aircraft and engines, M & R	75,000.00	—	75,000.00
Machines, general equipment and tools, M & R	336,536.00	60,000.00	396,536.00
Ordinance equipment for aircraft, M & R	14,250.00	—	14,250.00
Photographic equipment and supplies, M & R	75,000.00	3,000.00	78,000.00
Radio equipment and supplies for aircraft, M & R	—	3,000.00	3,000.00
Hydrogen, maintenance, equipment, operation of plants for production of	82,800.00	—	82,800.00
Test books, books of reference, and publications, purchase of	10,424.80	3,000.00	13,424.80
Maintenance & Repair of equipment, material, and instruments for Air Service schools	64,000.00	5,000.00	69,000.00
Purchase of equipment, material and instruments for use at Air Service schools	307,600.00	25,000.00	332,600.00
<b>Totals</b>	<b>\$4,432,734.00</b>	<b>\$1,126,000.00</b>	<b>\$5,558,734.00</b>

### MAINTENANCE & OPERATION OF STATIONS

	\$	\$	\$
Building and grounds at Air Service stations, M & R	330,000.00	—	330,000.00
Water	4,000.00	—	4,000.00
Light and power	30,000.00	—	30,000.00
<b>Totals</b>	<b>\$374,000.00</b>	<b>\$ —</b>	<b>\$374,000.00</b>

### RESEARCH, RESEARCH, RESEARCH & PRODUCTION

	\$	\$	\$
Purchase of land for development of	25,000.00	—	25,000.00
Lease of land for development of	900,000.00	—	900,000.00
Construction, maintenance, equipment and operation of plants	877,750.00	—	877,750.00
Experimentation with and exploration for	30,000.00	—	30,000.00
<b>Totals</b>	<b>\$1,852,750.00</b>	<b>\$ —</b>	<b>\$1,852,750.00</b>

### EXPERIMENTAL AND RESEARCH, ENGINEERING AND DEVELOPMENT

	Operation	Experimental Flight and Research	Total
Aircraft and accessories	\$ —	\$1,540,000.00	\$1,540,000.00
Radars, engines and accessories	—	1,250,000.00	1,250,000.00
Engines and accessories	—	550,000.00	550,000.00
Radio	—	18,000.00	18,000.00
<b>Totals</b>	<b>\$ —</b>	<b>\$3,358,000.00</b>	<b>\$3,358,000.00</b>

### PRODUCTION OF NEW AIRCRAFT, ENGINES AND ACCESSORIES

	\$	\$	\$
Special aviation clothing and similar equipment	30,000.00	—	30,000.00
Aircraft and their spare parts	17,784,050.00	2,000,000.00	19,784,050.00
Radars and their spare parts	442,700.00	—	442,700.00
Engines and their spare parts	3,181,000.00	—	3,181,000.00
Engines and their spare parts	—	3,375,000.00	3,375,000.00
Instruments and accessories for aircraft and engines	250,000.00	—	250,000.00
Ordinance equipment for aircraft	60,000.00	—	60,000.00
Radio equipment and supplies for aircraft	—	30,000.00	30,000.00
Machines, general equipment and tools	500,000.00	40,000.00	540,000.00
Photographic equipment and supplies for use in connection with aerial observation	413,000.00	50,000.00	463,000.00
<b>Totals</b>	<b>\$22,120,750.00</b>	<b>\$5,415,000.00</b>	<b>\$27,535,750.00</b>

### IMPROVEMENTS OF STATIONS, BANGERS, SHOPS AND GAS PLANTS; LANDING FIELDS

	\$	\$	\$
Gas plants, hangars and repair shops, construction of	80,000,000.00	—	80,000,000.00

### MISCELLANEOUS

	\$	\$	\$
Printing plants at Air Service stations, purchase of equipment and supplies for	9,000.00	25,000.00	34,000.00
Printing plants at Air Service Stations, M & R	—	4,000.00	4,000.00
Printing and bookbinding, Government Printing Office	—	15,000.00	15,000.00
Others expense while traveling by air	50,000.00	10,000.00	60,000.00
Others expense properly chargeable to Air Service appropriations	60,000.00	—	60,000.00
Damages, claims for not exceeding \$250.00	30,000.00	—	30,000.00
Relieving wrecked aircraft, supplies and accident in connection with	25,000.00	5,000.00	30,000.00
Office equipment, purchase and repair of	75,000.00	20,000.00	95,000.00
Subscriptions to foreign and professional periodicals and newspapers	8,000.00	—	8,000.00
Miscellaneous services not properly chargeable to other items	75,000.00	60,000.00	135,000.00
Miscellaneous supplies and equipment not properly chargeable to other items	215,000.00	20,000.00	235,000.00
Office salaries	33,750.00	—	33,750.00
Maneuvers, operations	100,000.00	—	100,000.00
<b>Totals</b>	<b>\$ 634,425.00</b>	<b>\$ 105,000.00</b>	<b>\$ 739,425.00</b>

### FUEL AND LUBRICANTS

	\$	\$	\$
Fuel and Lubricants	3,507,000.00	—	3,507,000.00
GRAVE TRAINS	\$48,700.00	\$11,200,000.00	\$11,248,700.00

# New Naval Fleet Development

These new types of ships for the United States Navy that will meet out the fleet and add tremendously to its effectiveness are now under construction.

The first of these, the new battleships, authorized in the 1915 building program, will be the largest and most powerful bat-

The *Lamphey* was also the pioneer ship of any Navy in its electrical propulsion field, her success establishing a precedent for future projects in the economical propulsion of ships of all classes. In refitting the *Lamphey* all of the coal handling gear has been removed and in its place will be created a flying



THE U. S. B. LAMPORT, AIRCRAFT CARRIER

tleships in the world. Next come the fleet submarines, three of which are being built at the Navy Yard, Portsmouth. These super-submarine craft will have sufficient speed and cruising radius to make it possible for them to accompany the fleet as a part of it.

The last of the line is the latest type of aircraft carrier, the *Lamphey*. This vessel is the old carrier *Jaguar*, refitted for her new role. Upward the *Lamphey* with the fleet will demonstrate the value of this type of vessel as she will be fitted with all the latest equipment for handling naval aircraft.

The *Jaguar* is now at the Navy Yard, Norfolk, undergoing alterations to refit her as an aircraft carrier. She has been renamed the *Lamphey* in honor of Professor Samuel Pierpont Lamport, whose extended passenger experiments on the problem of mechanical flight made his name widely known.



R. M. S. ARROW, BRATTLENE CARRIER

deck which will be located about 56 ft. above the waterline, extending from bow to stern, a length of about 525 ft. and with a width amidships of about 65 ft. This deck will be built at over 30 ft. to make an ideal platform for flying off and landing.

Masts will be provided on this deck for rigging as a steering device to facilitate landing. Catapults for projecting the planes to get them the necessary speed for flight will also be provided on this deck forward and aft.

An elevator will be installed for hoisting the planes from below to the flying deck and around this elevator, a platform will be provided to form a wind break for protection to the planes while being hoisted.

Two cranes with large outriggers, one on each side of the vessel, will be provided to hoist aircraft out of the water and hoist them on the landing deck, which is the deck next below the

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flying deck. Beneath the flying deck, traveling cranes will be provided for hoisting planes out of the hold and for transferring them fore and aft to the ship spaces and elevator. Shop facilities for repairing the planes will include machine shops, wing repairing shop, molding spaces, metal shop and various store rooms.

The hold spaces are being refitted for the storage of aircraft and their accessories, aircraft accessories including bombs and torpedoes, ship's accessories, fuel oil, gasoline and gun room spaces. An elaborate system will be provided for distributing gasoline and lubricating oils to various stations on the lower deck and flying deck.



LANDING DECK OF A BRATTLENE BRATTLENE CARRIER

The smoke pipe has been rearranged so as not to obstruct the flying deck and to keep the smoke clear of that deck, in order not to interfere with making a landing. This will be accomplished by providing a short smoke pipe on each side, clear of the deck. The funnels will be interconnected so that the smoke can always be discharged on the lee side. One of the smoke pipes will be arranged to hang downward when necessary necessary to discharge the smoke near the water; the other is arranged to discharge the smoke downward through a water spray.

A radio outfit will be provided, carried on masts, which are capable of being completely hoisted below the flying deck. An auxiliary radio outfit has also been provided so that the planes can be communicated with when the main radio outfit is hoisted.

The characteristics of the *Lamphey*, refitted as an aircraft carrier, are as follows:

Length between perpendiculars	525 ft.
Length overall	541 ft.
Beam	65 ft.
Deck area	12 acres
Displacement, normal	12,500 tons
Displacement, deck down	12,500 tons
Displacement, flying deck down	12,500 tons
Speed	45 knots
Range	4,000 miles

## Training for New Air Service Officers

Owing to the present shortage of commissioned personnel in the Army Air Service, and the consequent additional stress which devolves upon these officers and men to the fact that about eighty per cent, of the present commissioned strength of the Air Service is newly commissioned, only a small percentage of the new officers can be sent as students to the Special Service Schools. As U. S. No. 56, War Dept., requires all newly commissioned officers to be given the earliest possible opportunity to receive a basic military training, the difficulty must be partly overcome by arranging for instruction in flying as many of the newly commissioned officers as possible to pilot

and advanced flying schools where they will also receive in structure in basic military subjects.

These basic training will also be given to newly commissioned officers at all other Air Service special service schools. As it is not feasible that officers who are not students at special service schools be required to attend regular schools from which they could obtain knowledge or the knowledge of practical value, the fundamental study now to be met by some method of instruction other than a formal and perfunctory one, to be included in all commands, whereby officers may receive fundamental instruction so that they may not later be embarrassed or handicapped when called upon to perform satisfactory duties requiring the fundamental knowledge of various subjects.

Commanding officers of all Air Service activities have been directed to conduct such informal lecture and classes for newly commissioned officers as they may deem necessary, and such as will best accomplish the purposes desired. Air Service officers are also to be informed that they should have as their permanent military subjects for recitation reference, also copies of the following, which they may obtain free of charge upon request through the proper military channels in army area headquarters: Army Regulations, Compilation of Orders, Inspector's Regulations, Army Corps-Material, Non-Commissioned Officers' Manual, Field Service Regulations, War Department General Orders, Engineer's Field Manual, Rules of Land Warfare, and Small Arms Firing Regulations.





# International Aircraft Marking

By Ladislav D'Orey

The reconvening of Congress again brings to the fore the serious question of the international aerial convention which United States Ambassador Wallace signed on May 25, last, in Paris, with certain reservations.

Before this convention was read the United States to the other contracting states—twenty six in number, including our neighbors, Canada and Cuba—it must be ratified by Congress. After which, if the convention is not to remain a dead letter, there will be required an Act of Congress creating the necessary federal machinery which alone can provide for a nationwide enforcement of said convention.

That it is highly desirable for the United States to become a party to the Paris convention has been set forth in these columns before. The only unfortunate point about the convention is that the International Commission for Air Navigation, which it institutes in place of the Council of the League of Nations. This feature may defeat all attempts tending toward its ratification by Congress. If it does, then the next best thing Congress can be expected to do is to enact a national air legislation, which would be based on the text of the Paris convention except for such provisions of the latter as were reserved by Ambassador Wallace and for its connection with the League of Nations.

Such national air legislation should reproduce the agreed provisions of the Paris convention as closely as possible, for it is of the utmost importance for American aviation that a single set of regulations apply to air navigation throughout the United States as well as in Canada, Mexico and Cuba. Our nearest four neighbors mutually agreed on the identical field for aerial intercourse. American mail and passenger air lines are already operating from the extreme tips of this country to Canada and Cuban ports, and only the other day two airplanes flew without such aid from London, New, to Mexico City.

There are but small beginnings. Larger enterprises have so far failed to materialize mainly because of the legal insecurity American aircraft have suffered from, owing to the absence of consistent national legislation. Once the latter exists it cannot be doubted that American aviation will derive the same benefits from a single national air code as has been the case with the countries that possess such an instrument of control.

## Nationality and Registration Marks for Aircraft

The demand for a national air code undoubtedly brings up the question of the nationality and registration marks for air craft. Such marks must be capable of prompt identification from the ground when aircraft are in flight, so that trespassers of the aerial law may be quickly apprehended. The reports which framed the Paris convention very carefully considered the point and agreed that the nationality and registration marks should entirely consist of letters, rather than of letters and numerals. Annex A of the convention contains the following provisions as to the marking of aircraft:

The nationality mark shall be represented by a single letter, while the registration mark shall be represented by a group of four letters, each group containing at least one vowel. For this purpose the letter "V" shall be considered as a vowel.

The complete group of five letters shall be used as a tail sign of the particular aircraft in making or receiving signals by wireless telegraphy.

Private machines—that is, airplanes, seaplanes and dirigibles—shall be marked in the following manner:

The nationality and registration marks, divided by a hyphen, shall be painted in black on a white ground on (1) the upper surface of the top main plane, (2) the lower surface of the bottom main plane, and (3) the right and left sides of the fuselage or of the hull, between the main planes and the tail plane.

The nationality mark alone shall be painted on short propellers.

And on (1) the right and left sides of the saddle, or on its outer sides of the outer rudders if more than one is fitted and (2) the right and left sides of the upper and lower surfaces of the horizontal stabilizer or of the elevator, whichever is the larger.

The height of the marks on the main planes and tail plane shall be equal to four-fifths of the chord, and in case of the rudders shall be as large as possible. The width of the letters shall be two-thirds of their height, the thickness one-sixth of their height. The letters shall be painted in plain block letters.

Fig. 1 illustrates a Canadian seaplane marked in accordance with these provisions, except that no mark appears on the upper surface of the top plane. Figs. 2 and 3 show two variously marked machines of French and Polish nationality, respectively. On the F-PRAS the marks are painted directly on the machine instead of being displayed on a white background, as a mark they are hard to catch. At an altitude of any 1,000 feet these marks would hardly be visible from the ground, whereas those of the Canadian seaplane would all show up clearly. Furthermore, the nationality mark on its machine is placed too low to be easily spotted. The marking of third machines, the L-ANTE, is even worse. The letters painted on the fuselage are far below the ones prescribed by the convention and the rudder mark is totally missing.

The main object of this criticism is to show how important it is to carry out the provisions of a regulation correctly if the latter is to be of any utility at all. The rapid rate of trend of aircraft makes such compliance particularly imperative.

## An Appeal to Imagination

But the visible display of aircraft marks is not the only requirement desirable from the viewpoint of prompt identification. An aircraft mark may be visible enough, but if the nationality letter does not really convey the name of the country it stands for, its utility will be greatly curtailed. The problem is complicated by the fact that while there are but twenty-six letters in the alphabet, the sovereign states of the world number about eighty. It follows that some countries have to be designated by two letters.

The system adopted in the convention follows this principle by assigning to the great powers, which are the world's principal aircraft users, a single letter to denote their nationality, while all the other contracting states are designated by two letters. In the case of the latter, therefore, the registration mark (group of four letters) always begins with a given letter, fixed by the convention.

It would be interesting to know what reference served as its selection of some of the letters the convention provides for a few countries: "W" for Great Britain, "F" for France, "U" for Italy, "J" for Japan, which were legal, the United States is designated by the letter "N". Why "N" should be the nationality sign of this country is a question which equally has yet failed to explain. Surely, this is not a characteristic letter for designating the United States, the use of an "N" on an aircraft is given the letters "U" or "A".

The method followed in assigning the two-letter nationality marks is even more surprising. It would have been expected that where the names of two or more countries began with the same letter, that would be retained as the common initial, while the second letter would be specially selected with a view to distinguishing the names from one another.

For instance, "B-B" would stand for Belgium, "B-B" for Bolivia, "B-B" for Brazil, and so forth. Instead the Convention assigns to each country the second letter in common, while the first is chosen arbitrarily. Actually "D-D" stands for Belgium, "C-C" for Bolivia, and "B-B" for Brazil, which seems to say that on the tailplane of an airplane the letter "B"

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Fig. 1



Fig. 2



Fig. 3





and if these points can be measured in terms of the incident radiation, then the fraction absorbed can be found. For example, if a fraction  $R$  of the incident radiant energy is absorbed, a fraction  $P$  is transmitted, and a fraction  $A$  is absorbed, the following relation is obvious:

$$R + P + A = 1$$

or  $A = 1 - (R + P)$

Thus the absorption is given in terms of the transmission and reflection, and the quantity  $A$  times the incident energy represents the rate at which heat will be produced in the fabric. A fraction of this heat will be lost by conduction and convection, while another portion will be reradiated. The portion reradiated depends upon a radiance property of the

surface versus about the same for each of the fabrics, and the measurements give a reliable comparison of the different fabrics. Table No. 1 shows the results.

#### Temperature of Balloons

The temperature and the corresponding lifting power of the gas are determined, solely by the temperature of the fabric, for the absorption of radiation by the gas is negligible.

The temperature of a thin sheet of balloon fabric upon which radiation is falling, is dependent on (1) the intensity of the radiation, (2) the absorbing power of the surface for the incident radiation, (3) the emissivity of the surface at the temperature in question; (4) the loss of heat by conduction

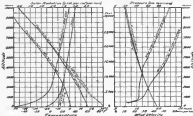


FIG. 3

radiant, called the emissivity. The magnitude and interrelations of these factors determine the temperature rise of the fabric.

The general method for determining the reflecting power of any surface is to measure the energy (radiant power) of a beam of light before and after reflection from the surface. The ratio of these values gives the reflecting power for the light used as a source. The apparatus used for making these measurements were devised by Dr. W. W. Coblentz, of the Bureau, and we are indebted to him for the loan of the original apparatus and has assistance in making the preliminary measurements. The method is brief was to measure the energy of a uniform beam of light by means of a bolometer. The energy of the beam was then measured after reflection at several angles from a sample of the fabric under test. The transmission was determined with the same apparatus with one modification, the method was to measure the energy of the beam of sunlight before and after passage through the fabric.

Because of the difficulty of using a moving body, such as the mirror in a source of radiation for measurements where a constant interval of time existed between the measurements of incident and reflected light, solar radiation was used in only a few of the measurements. A fair substitute for the sun was found by using the radiation from a reflector-diffused tungsten lamp transmitted by a 2-percent solution of cuprous chloride, 0.5 cm. in thickness.

While the spectral distribution of energy in the radiation from the lamp after passage through cuprous chloride solution is somewhat similar to that of sunlight, yet it is not as rich in the shorter wave lengths. The solar radiation reaching the earth's surface, as shown in Fig. 1, has a maximum intensity at about 0.5 microns, while that of the radiation in the reflector and transmission measurements has a maximum near 0.6 microns.

It was noted that without exception, when measurements were made, the reflecting power for sunlight was somewhat less than for the light used. This was also true for the transmitting power. However, the reflectance and trans-

mission, both from the front and back surfaces. An additional mathematical analysis of these features is given in T. P. 125 but will not be accepted here.

It was decided to make a preliminary survey of the problem by means of measurements on a small-sized model balloon. These measurements were designed to show the temperature of the surface of the balloon at different points and the temperature gradient in the gas contained in the envelope.

The model balloon which was used for the measurements of the temperature of the fabric and gas was a model of a streamline drogue and had a maximum diameter of 3 feet and a length of 12 feet. The measurements of temperature were made by means of upper-extremity thermocouples and a potentiometer. The couples were made by soldering the elements in the form of No. 36 B. & S. gage wire, to very thin, light metal covers, 1.5 mm. by 3.0 mm. Some of these receivers were in contact with the outer surface of the balloon by means of Canada balsam. In this way a cross-fiber contact was made between the thermocouple and the fabric and yet the receiver was not so massive as to affect the temperature measurements. Similar thermocouples of the same size were supported at various points inside the balloon by means of the No. 36 wire leads. These could be drawn from one side of the balloon to the other, thus exploring the temperature of the gas throughout, a slender cross section of the balloon.

In order to determine the relative rise in temperature of different fabrics, seven samples T, on, on a side were supported in each side of the balloon. This condition was found in every series of measurements made, and it can be considered, therefore, constant. The higher temperatures at the lower surfaces is probably due to a greater intensity of transmitted radiation at this point. The lowest temperatures measured on this model balloon was 66.6 deg. C (152 deg. F) with a surrounding air temperature of 20.6 deg. C (70 deg. F) and a solar radiation intensity of about 1.42 g.

In order to place results obtained at different times and under different conditions of solar radiation and air temperature on a comparable basis, we have assumed a standard condition where the air temperature is 25 deg. C, and the solar intensity equals 1.4 g. calories per square centimeter per minute.

The temperature of the surface of the balloon and of the contained gas at various points is shown in Fig. 3. In this figure the locations of the thermocouples attached to the fabric and inside the balloon are shown by black dots and the temperature in degrees Centigrade at each point is indicated by the figures.

TABLE 1—MEASUREMENT CHARACTERISTICS OF BALLOON FABRICS AND THEIR INCREASE IN TEMPERATURE WHEN EXPOSED TO SUNLIGHT (Except where marked, all fabrics are of equal composition with an average density of 0.0015 of the chlorine contained)

No.	Description of Fabric	Sample at fabric	Radiation from Thermocouple			Radiation from Thermocouple		
			Reflected	Transmitted	Absorbed	Reflected	Transmitted	Absorbed
			at 25° C	at 25° C	at 25° C	at 25° C	at 25° C	at 25° C
1	Acrylic fabric, silver, 40 days exposure, almost black		14.9	0.0	42.8	22.1		
2	Acrylic fabric, aluminum coated		30.7	0.0	40.0			
3	Experimental acrylic fabric, aluminum coated		40.0	0.0	40.1	38.0		
4	Experimental acrylic fabric, aluminum coated		41.0	0.0	39.1	38.0		
5	Experimental acrylic fabric, aluminum coated		40.2	0.0	39.0	38.0		
6	Experimental acrylic fabric, aluminum coated		39.4	1.4	38.0			
7	Experimental acrylic fabric, aluminum coated		32.6	0.4	37.0			
8	Experimental acrylic fabric, aluminum coated		40.0	1.0	38.0	38.0		
9	Experimental acrylic fabric, aluminum coated		30.0	12.4	38.4			
10	Experimental acrylic fabric, aluminum coated		40.0	0.1	38.0	38.0		
11	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
12	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
13	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
14	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
15	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
16	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
17	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
18	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
19	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
20	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
21	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
22	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
23	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
24	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
25	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
26	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
27	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
28	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
29	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
30	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
31	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
32	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
33	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
34	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
35	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
36	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
37	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
38	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
39	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
40	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
41	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
42	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
43	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
44	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
45	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
46	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
47	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
48	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
49	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
50	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
51	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
52	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
53	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
54	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
55	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
56	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
57	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
58	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
59	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
60	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
61	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
62	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
63	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
64	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
65	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
66	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
67	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
68	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
69	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
70	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
71	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
72	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
73	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
74	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
75	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
76	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
77	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
78	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
79	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
80	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
81	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
82	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
83	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
84	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
85	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
86	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
87	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
88	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
89	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
90	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
91	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
92	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
93	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
94	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
95	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
96	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
97	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
98	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
99	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		
100	Experimental acrylic fabric, aluminum coated		30.1	3.0	37.0	38.0		

A significant fact to be noted in these figures is that the maximum temperature occurs just below the beginning of the balloon in each side of the balloon. This condition was found in every series of measurements made, and it can be considered, therefore, constant. The higher temperatures at the lower surfaces is probably due to a greater intensity of transmitted radiation at this point. The lowest temperatures measured on this model balloon was 66.6 deg. C (152 deg. F) with a surrounding air temperature of 20.6 deg. C (70 deg. F) and a solar radiation intensity of about 1.42 g.

It should be noted that a difference in temperature of as much as 25 deg. C may exist in different parts of the gas.

It is therefore important, in reporting such measurements to specify the location of the point at which the measurements are made. The pressure of the balloons as the full sun envelope would probably cause the upper and lower surfaces to expand about the same, but the difference in expansion between the upper and lower surfaces. The balloon acting as a sort of diaphragm across the interior of the balloon would absorb the transmitted and reflected energy from the sun and heat the gas in the upper surface by radiation, just as did the lower surface in the model balloon. The temperature would consequently be relatively cooler than in the case of the full sun envelope. It is worthy while, also, to note the regularity with which the temperature changes at different points as a balloon when passing from direct sunlight into the shade or from one sun to another. For this purpose measurements of the temperature of the upper and lower surfaces and the temperature of the gas at the center and half way between the center and upper surface, were made, nearly simultaneously, as far as possible. These measurements were made on the model balloon at a time when small dense clouds were visible. The results are shown graphically in Fig. 4. The solid line represents the true observed temperature and the dashed line the true interval in minutes. The curves A, B, C, and D represent, respectively, the temperature of the upper surface of the gas midway between the water and the top, of the gas at the center, and the temperature of the lower surface. With rapidly changing temperature the temperature of the gas at the center is seen to be about wholly dependent on that of the lower surface and to lag behind it from 1 to 2 minutes.

During the time interval (A—B) 3 minutes the temperature changes were as follows:



If the "upper" gas temperature represents an average for the upper half of the balloon and a mean of the center gas and lower surface temperatures as average data from the lower half, then the temperature change of the gas is 8.4 deg. C. The actual temperature was 65 deg. C or 318 deg. F. Also, before, the temperature of the gas was 20.6 deg. C (70 deg. F). In addition, the pressure remained constant. This means a volume, or buoyancy, change of 0.9 per cent per minute. To show the effect of such a rapid change in temperature, the pressure was measured, that the average temperature of the gas in an interval of 100/1000 cubic feet rapidly within such a change. In such a case it could be necessary to 100 cubic feet of gas at 1000 cubic feet per minute to keep the envelope properly inflated. It would be interesting to observe the actual magnitude of these changes in practice.

The rise in temperature could be measured for all fabrics under ordinary conditions, if a fair comparison is to be secured. It was found that the increase in temperature shows that of the surrounding air was directly proportional to the solar intensity over a rather wide range. It was therefore possible to make the temperature measurements for various solar intensities and extrapolate for a standard solar intensity which was taken 1.4 g. calories per square centimeter per minute. All results are given for 1000 air surrounding the fabric at a temperature of 25 deg. C, which is the temperature at which permeability measurements of the fabrics under test were made. The solar intensity range for a number of fabrics, which measured in this manner is shown in the last column of Table 1.

#### Selection of Fabric

There are so many sections which may be desirable in a balloon fabric that the requirements for each may be so conflicting that the resultant choice may be a compromise. The choice will, however, take into consideration the specific use

for which the fabric is intended. The considerations of strength and elastic properties are more or less independent of the other properties of the fabric. Weight, radiation characteristics, color, visibility, and durability are, however, all interrelated and must be considered together.

It has been pointed out that from the standpoint of the

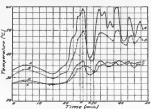


FIG. 4

effect of radiation in raising the temperature, a nearly translucent fabric or a perfectly reflecting surface would be most desirable. Such a fabric is not a possibility at the present time, however. A fabric with a white rubber coating accompanied with fine scale would also be desirable from this standpoint, although not far from others. Fabrics Nos. 8 and 9 in Table I, which have the color of the unadorned cloth and the pale yellow of pure vulcanized rubber, show a very small increase in temperature, but because they lack any protection against the injurious action of light their useful life would be very short. This objection, however, does not apply to aluminum-coated fabrics. The aluminum coating not only protects the pre-existing rubber film by virtue of its opacity to light, but also, because of its high reflecting power, remains comparatively cool when exposed to sunlight. The deterioration of fabrics 8 and 9 when exposed to the weather in use at Washington is about ten times as rapid as in the case of the best aluminum-coated fabrics.

For military purposes where low visibility is desirable, there are numerous advantages to the use of the lightest and highly reflecting aluminum coat. Also, the weight of the aluminum coating may be undesirable for airplanes where a very light fabric is required. In these cases it is necessary to use a dyed or pigmented rubber-coated fabric which meets the color and visibility requirements and yet shows a minimum temperature effect and satisfactory durability characteristics.

It has been possible only to allude to these few paragraphs the many considerations which attend the choice of a fabric for aircraft. The effect of radiation upon the fabric should not be lost sight of in any case, even if it is not of great importance, frequently as improvement of the fabric in this respect can be obtained without the sacrifice of any other desirable characteristics.

#### Standard Charges for Airport Service

The following tables are reproduced from the French journal *L'Aviation* and show the standard charges for the various services at licensed airports. It will be recalled that the terms of the International Air Convention require that the schedule be uniform in any country and that no distinction be made between foreign and national aircraft. Another regulation provides that the schedule be posted in a conspicuous place at the airport.

Loading Charges	FRANCE			BELGIUM		
	A	B	C	A	B	C
Suburban Aircraft, (at the common shelter with the right to personal baggage)	3.00	4.00	6.00	3.00	4.00	6.00
Aircraft of personnel (from the right to personal baggage)	3.00	4.00	6.00	3.00	4.00	6.00
Transit Aircraft (right to personnel)	15.00	20.00	30.00	12.00	16.00	24.00

Rate for holders of books of tickets	FRANCE			BELGIUM		
	A	B	C	A	B	C
Ordinary rate	6.00	10.00	30.00	6.00	10.00	30.00

George Charges (24 Hours)	FRANCE			BELGIUM		
	A	B	C	A	B	C

Transit Aircraft (right to personnel)	20.00	25.00	35.00	13.00	18.00	28.00
Suburban Aircraft (right to personnel)	8.00	10.00	15.00	8.00	10.00	15.00
Personal Term:						

This is on the basis of a minimum of three months at the rate of 1 fr per square meter per month. No right to personnel and lighting etc. charges paid by law.

Transit Aircraft	FRANCE			BELGIUM		
	A	B	C	A	B	C
Transit Aircraft	12.00	20.00	30.00	12.00	20.00	30.00
Personal Term:	6.00	10.00	15.00	6.00	10.00	15.00

In France and Belgium, Class A contains machines of up to 1000 cc, Class B, from 1000 to 2000 cc, Class C, from 2000 to 3000 cc, Class D, from 3000 to 4000 cc, Class E, from 4000 to 5000 cc, Class F, from 5000 to 6000 cc, Class G, from 6000 to 7000 cc, Class H, from 7000 to 8000 cc, Class I, from 8000 to 9000 cc, Class J, from 9000 to 10000 cc, Class K, from 10000 to 11000 cc, Class L, from 11000 to 12000 cc, Class M, from 12000 to 13000 cc, Class N, from 13000 to 14000 cc, Class O, from 14000 to 15000 cc, Class P, from 15000 to 16000 cc, Class Q, from 16000 to 17000 cc, Class R, from 17000 to 18000 cc, Class S, from 18000 to 19000 cc, Class T, from 19000 to 20000 cc, Class U, from 20000 to 21000 cc, Class V, from 21000 to 22000 cc, Class W, from 22000 to 23000 cc, Class X, from 23000 to 24000 cc, Class Y, from 24000 to 25000 cc, Class Z, from 25000 to 26000 cc, Class AA, from 26000 to 27000 cc, Class AB, from 27000 to 28000 cc, Class AC, from 28000 to 29000 cc, Class AD, from 29000 to 30000 cc, Class AE, from 30000 to 31000 cc, Class AF, from 31000 to 32000 cc, Class AG, from 32000 to 33000 cc, Class AH, from 33000 to 34000 cc, Class AI, from 34000 to 35000 cc, Class AJ, from 35000 to 36000 cc, Class AK, from 36000 to 37000 cc, Class AL, from 37000 to 38000 cc, Class AM, from 38000 to 39000 cc, Class AN, from 39000 to 40000 cc, Class AO, from 40000 to 41000 cc, Class AP, from 41000 to 42000 cc, Class AQ, from 42000 to 43000 cc, Class AR, from 43000 to 44000 cc, Class AS, from 44000 to 45000 cc, Class AT, from 45000 to 46000 cc, Class AU, from 46000 to 47000 cc, Class AV, from 47000 to 48000 cc, Class AW, from 48000 to 49000 cc, Class AX, from 49000 to 50000 cc, Class AY, from 50000 to 51000 cc, Class AZ, from 51000 to 52000 cc, Class BA, from 52000 to 53000 cc, Class BB, from 53000 to 54000 cc, Class BC, from 54000 to 55000 cc, Class BD, from 55000 to 56000 cc, Class BE, from 56000 to 57000 cc, Class BF, from 57000 to 58000 cc, Class BG, from 58000 to 59000 cc, Class BH, from 59000 to 60000 cc, Class BI, from 60000 to 61000 cc, Class BJ, from 61000 to 62000 cc, Class BK, from 62000 to 63000 cc, Class BL, from 63000 to 64000 cc, Class BM, from 64000 to 65000 cc, Class BN, from 65000 to 66000 cc, Class BO, from 66000 to 67000 cc, Class BP, from 67000 to 68000 cc, Class BQ, from 68000 to 69000 cc, Class BR, from 69000 to 70000 cc, Class BS, from 70000 to 71000 cc, Class BT, from 71000 to 72000 cc, Class BU, from 72000 to 73000 cc, Class BV, from 73000 to 74000 cc, Class BW, from 74000 to 75000 cc, Class BX, from 75000 to 76000 cc, Class BY, from 76000 to 77000 cc, Class BZ, from 77000 to 78000 cc, Class CA, from 78000 to 79000 cc, Class CB, from 79000 to 80000 cc, Class CC, from 80000 to 81000 cc, Class CD, from 81000 to 82000 cc, Class CE, from 82000 to 83000 cc, Class CF, from 83000 to 84000 cc, Class CG, from 84000 to 85000 cc, Class CH, from 85000 to 86000 cc, Class CI, from 86000 to 87000 cc, Class CJ, from 87000 to 88000 cc, Class CK, from 88000 to 89000 cc, Class CL, from 89000 to 90000 cc, Class CM, from 90000 to 91000 cc, Class CN, from 91000 to 92000 cc, Class CO, from 92000 to 93000 cc, Class CP, from 93000 to 94000 cc, Class CQ, from 94000 to 95000 cc, Class CR, from 95000 to 96000 cc, Class CS, from 96000 to 97000 cc, Class CT, from 97000 to 98000 cc, Class CU, from 98000 to 99000 cc, Class CV, from 99000 to 100000 cc, Class CW, from 100000 to 101000 cc, Class CX, from 101000 to 102000 cc, Class CY, from 102000 to 103000 cc, Class CZ, from 103000 to 104000 cc, Class DA, from 104000 to 105000 cc, Class DB, from 105000 to 106000 cc, Class DC, from 106000 to 107000 cc, Class DD, from 107000 to 108000 cc, Class DE, from 108000 to 109000 cc, Class DF, from 109000 to 110000 cc, Class DG, from 110000 to 111000 cc, Class DH, from 111000 to 112000 cc, Class DI, from 112000 to 113000 cc, Class DJ, from 113000 to 114000 cc, Class DK, from 114000 to 115000 cc, Class DL, from 115000 to 116000 cc, Class DM, from 116000 to 117000 cc, Class DN, from 117000 to 118000 cc, Class DO, from 118000 to 119000 cc, Class DP, from 119000 to 120000 cc, Class DQ, from 120000 to 121000 cc, Class DR, from 121000 to 122000 cc, Class DS, from 122000 to 123000 cc, Class DT, from 123000 to 124000 cc, Class DU, from 124000 to 125000 cc, Class DV, from 125000 to 126000 cc, Class DW, from 126000 to 127000 cc, Class DX, from 127000 to 128000 cc, Class DY, from 128000 to 129000 cc, Class DZ, from 129000 to 130000 cc, Class EA, from 130000 to 131000 cc, Class EB, from 131000 to 132000 cc, Class EC, from 132000 to 133000 cc, Class ED, from 133000 to 134000 cc, Class EE, from 134000 to 135000 cc, Class EF, from 135000 to 136000 cc, Class EG, from 136000 to 137000 cc, Class EH, from 137000 to 138000 cc, Class EI, from 138000 to 139000 cc, Class EJ, from 139000 to 140000 cc, Class EK, from 140000 to 141000 cc, Class EL, from 141000 to 142000 cc, Class EM, from 142000 to 143000 cc, Class EN, from 143000 to 144000 cc, Class EO, from 144000 to 145000 cc, Class EP, from 145000 to 146000 cc, Class EQ, from 146000 to 147000 cc, Class ER, from 147000 to 148000 cc, Class ES, from 148000 to 149000 cc, Class ET, from 149000 to 150000 cc, Class EU, from 150000 to 151000 cc, Class EV, from 151000 to 152000 cc, Class EW, from 152000 to 153000 cc, Class EX, from 153000 to 154000 cc, Class EY, from 154000 to 155000 cc, Class EZ, from 155000 to 156000 cc, Class FA, from 156000 to 157000 cc, Class FB, from 157000 to 158000 cc, Class FC, from 158000 to 159000 cc, Class FD, from 159000 to 160000 cc, Class FE, from 160000 to 161000 cc, Class FF, from 161000 to 162000 cc, Class FG, from 162000 to 163000 cc, Class FH, from 163000 to 164000 cc, Class FI, from 164000 to 165000 cc, Class FJ, from 165000 to 166000 cc, Class FK, from 166000 to 167000 cc, Class FL, from 167000 to 168000 cc, Class FM, from 168000 to 169000 cc, Class FN, from 169000 to 170000 cc, Class FO, from 170000 to 171000 cc, Class FP, from 171000 to 172000 cc, Class FQ, from 172000 to 173000 cc, Class FR, from 173000 to 174000 cc, Class FS, from 174000 to 175000 cc, Class FT, from 175000 to 176000 cc, Class FU, from 176000 to 177000 cc, Class FV, from 177000 to 178000 cc, Class FW, from 178000 to 179000 cc, Class FX, from 179000 to 180000 cc, Class FY, from 180000 to 181000 cc, Class FZ, from 181000 to 182000 cc, Class GA, from 182000 to 183000 cc, Class GB, from 183000 to 184000 cc, Class GC, from 184000 to 185000 cc, Class GD, from 185000 to 186000 cc, Class GE, from 186000 to 187000 cc, Class GF, from 187000 to 188000 cc, Class GG, from 188000 to 189000 cc, Class GH, from 189000 to 190000 cc, Class GI, from 190000 to 191000 cc, Class GJ, from 191000 to 192000 cc, Class GK, from 192000 to 193000 cc, Class GL, from 193000 to 194000 cc, Class GM, from 194000 to 195000 cc, Class GN, from 195000 to 196000 cc, Class GO, from 196000 to 197000 cc, Class GP, from 197000 to 198000 cc, Class GQ, from 198000 to 199000 cc, Class GR, from 199000 to 200000 cc, Class GS, from 200000 to 201000 cc, Class GT, from 201000 to 202000 cc, Class GU, from 202000 to 203000 cc, Class GV, from 203000 to 204000 cc, Class GW, from 204000 to 205000 cc, Class GX, from 205000 to 206000 cc, Class GY, from 206000 to 207000 cc, Class GZ, from 207000 to 208000 cc, Class HA, from 208000 to 209000 cc, Class HB, from 209000 to 210000 cc, Class HC, from 210000 to 211000 cc, Class HD, from 211000 to 212000 cc, Class HE, from 212000 to 213000 cc, Class HF, from 213000 to 214000 cc, Class HG, from 214000 to 215000 cc, Class HH, from 215000 to 216000 cc, Class HI, from 216000 to 217000 cc, Class HJ, from 217000 to 218000 cc, Class HK, from 218000 to 219000 cc, Class HL, from 219000 to 220000 cc, Class HM, from 220000 to 221000 cc, Class HN, from 221000 to 222000 cc, Class HO, from 222000 to 223000 cc, Class HP, from 223000 to 224000 cc, Class HQ, from 224000 to 225000 cc, Class HR, from 225000 to 226000 cc, Class HS, from 226000 to 227000 cc, Class HT, from 227000 to 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477000 to 478000 cc, Class RL,

incorporation of any changes which appear necessary or advisable, from the time of the first two. No payment will be made to the contractor for any change or alteration in any airplane which does not affect the basic design and which is made in any uncompleted aircraft during any of the periods during which the contractor is to be paid in whole or in part. The cost of such other changes and alterations as are ordered by the government in the airplane, will be covered by an agreement supplemental to the contract, upon a fixed-price basis.

(c) That the contractor must also furnish a final analysis of design, a complete set of experimental working drawings and a bill of material, all of the third airplane, to be in the form of Van Dikes. These shall be delivered within sixty days after the delivery of the third airplane. A complete set of experimental working drawings is defined as those drawings from which the third airplane has been constructed. They must embrace the dimensions and changes made in that airplane and include no obsolete data. Sketches will not suffice and drawings to a definite scale must be provided. These working drawings need not be detailed drawings in the sense that they must give a detail of every part of the airplane, but should consist of approximately 500 to 600 separate parts from which an experimental construction of airplane could reproduce the airplane in limited quantities.

(d) The government will furnish, without cost to the contractor, the instruments, instruments and standard accessories mentioned in the specifications, and will endeavor to furnish such raw materials as are not readily available, at prices approximately equal to the market value.

(e) That partial payments will be made to the contractor as the work progresses.

(f) That the contract must be accompanied by an approved master plan in duplicate for 50 per cent of the amount of the contract.

(g) That all airplanes constructed for test be delivered in good condition by the contractor, i.e., the headquarters of the Engineering Division, Air Service, where each will be tested by this Division and accepted or rejected according to the terms of the contract. The specifications prescribe the tests which will allow the contractor an approval of the airplane, within which is readily and easily any rejected airplane, with a view to making the same acceptable to the government.

(h) That the government may terminate the contract at any time by suspending the contractor in accordance with the standard provisions of Engineering Division contracts.

#### Japanese Air Mail Progress

The Japanese government has completed all arrangements for the early inauguration of a comprehensive air mail service throughout Japan. With characteristic efficiency and determination to detail official reports have been prepared dealing with every phase of the aviation industry in Japan as well as schedules relative to pilots, aerodromes and suitable landing airports. It is possible to carry four passengers in addition to mail and up to 600 lb. of freight are said to be recommended for adoption by Mr. Aoyama of the Japanese Transport Department who was mentioned to state that European methods of commercial aviation. Tokyo and Osaka are expected to be the first cities to be put into regular commercial service by air and the Japanese Imperial Aviation Society is to be authorized with the inauguration of all the new air routes.

#### Post Field to Fabricate Complete Spherical Balloon

The Ballon and Airship personnel of the U. S. Air Service at Post Field, Fort Worth, Okla., has planned to fabricate a complete spherical balloon. The work is to be done by the enlisted men of the field under the direction of the officers. The balloon will be made of salvaged material. The men will select the fabric, cut them, cut them, cut them, and assemble them in the finished balloon. Further than this it is planned to make a net for the balloon with all the different of rigging—essentially which that outside.

This balloon is expected to fly in regular five balloon service. The men who run it, but know, knowing that it is a product of his work and that of his personnel will be very appreciative with a thorough knowledge of essential details means.

#### Aeronautical Progress in Indiana

In the past year there has been a great public school interest movement in Indiana as regard to aeronautics, started by the Curtiss Indiana Co. of Kokomo, and as many as twenty thousand persons have had rides in airplanes which this company has sent broadcast over the State. The company was organized primarily to promote the distribution of Curtiss airplanes, and includes an air line of aircraft, Edward Hanna of the Indiana Automobile Co., and many prominent Kokomo men. The company is under the direct management of Lieut. W. M. Pagley.

Besides a fine flying field and hangars, near the Kokomo County Club, the company has established a service department handling a complete line of Curtiss parts. It also has organized a system of flying clubs or clubs throughout the state, which enables the person of average income and ordinary physique to become pilot under an airplane hangar and to receive thorough flying instruction.

#### Aero Club of New England Elections

The Aero Club of New England held the monthly meeting at the City Club, Boston November 23, with President Godfrey L. Cabot presiding and about 30 members and guests attending. Rear Admiral Stanley S. Robinson, commandant at the Boston Navy Yard, and Maj. Gen. David C. Sharpe, commandant of the 1st Army Corps area, talked on development of aviation in the military and naval service. Parker H. Kinsale, a director of the club, gave an illustrated talk.

The following officers were elected: Godfrey L. Cabot, president; Joseph S. Hathaway and J. Walter Flagg, vice presidents; William Carroll Hall, secretary; Alfred H. Shugley, treasurer; Godfrey L. Cabot, J. Walter Flagg, Charles J. Giddies, Joseph S. Hathaway, William Carroll Hall, Henry Howard, Alfred H. Shugley, Nelson H. Smith, Carl E. Stearns, John J. Van Vleet, and Parker H. Kinsale and Cyrus H. Russell, directors.

#### Spokane News

The newest development in aviation in the Spokane territory is a plan to establish commercial flight between Spokane and Seattle. The United States Airways Corporation is behind the plan. It reports to use rebuilt DeSoto cars equipped with 500 hp. Liberty engines. Each plane will have a carrying capacity of six or seven passengers. The company has 34 people enrolled at its aviation school, one being a Japanese ordered by the government of Japan.

The company intends to develop aerial photography and make trips over the national parks, taking aerial pictures while flying. Refueling will be made through the Pacific coastway. Arrangements for service have already been made. A service of photographing cities or making views of any commercial concern's place, or pictorial views of land will be available.

#### German Aero Clubs Combine

The Aero clubs of Karlsruhe, Gross Tschernau and Sauerbrunn have combined to form a national aviation society. The national meeting took place at Garmisch in the Prussian, and was largely attended by delegates from many district societies and representatives of important manufacturing and commercial interests. A resolution was unanimously adopted pledging the members to prevent by all possible means the treatment of foreign capital in German air transport companies. The headquarters of the new society are at Garmisch.

#### Greenboro N. C. New York Flight

Pilot Charles W. Meyers of East Orange, N. J., recently made a mid-air flight from Greenboro, N. C., to New York. Distance of slightly over 300 miles in 4 1/2 hours and 30 min. flying time. He used a Caudron Curtiss equipped with an OX-5 motor, and carried in a passenger John Davis of Greenboro. The flight was the first of a series of flights planned for the State of Alabama, Georgia, South Carolina and North Carolina.



## —AND NOW EVEN ITS NAME IS AMERICAN!

WITH the many mechanical changes and improvements in design now effective the original American-made Hispano-Suiza Engine becomes in name as well as in effect a purely American product and will henceforth be called the Wright Aeronautical Engine.

The original conception of the Hispano screw cylinder sleeve is substantially the only remaining foreign-conceived part of the design

and for several years all Wright-built models of this most successful engine have been of American design right up to improvements over the original foreign model.

Such an outstanding American achievement deserves the typically American name of this Company which has stood and still stands for all that is best in Aeronautical engineering.

## WRIGHT AERONAUTICAL CORPORATION

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# W R I G H T

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*Against the Following Risks*

1. FIRE AND TRANSPORTATION.
2. THEFT (Of the machine or any of its parts).
3. COLLISION (Damage sustained to the plane itself).
4. PROPERTY DAMAGE (Damage to the property of others).

### SPECIAL HAZARDS

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AGENTS IN CITIES, TOWNS AND VILLAGES THROUGHOUT THE UNITED STATES AND ITS POSSESSIONS, AND IN CANADA, MEXICO, CUBA, PORTO RICO AND CENTRAL AMERICA.

*Aircraft, Automobiles, Fire and Lightning, Explosion, Rail, Marine (Inland and Ocean), Fossil Fuel, Pockets and Combinations, Registered Mail, Buses, Rental Values, Rest and Clerk Commission, Sprinkler Leakage, Tourist's Baggage, Life and Occupancy, Windscreens.*

STRENGTH

REPUTATION

SERVICE

*"You can imagine our wild delight as we left gaming world finally we passed . . . about sun-down and were four miles ahead when night came, grey black."*

Here is a paragraph from a tale that a pilot told. He describes his part in the recent International Balloon Race, from Birmingham, Alabama, north. He pictures vividly the thrills of this peace-time adventure, when men adrift under gas bags strove for mileage as-the-crow-flies. To win, they were dependent upon great good judgment, upon ballast and upon the will of the winds.

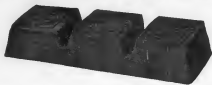
There are Goodyear Spherical Balloons ready for immediate delivery; they are well made, suitable for sport lovers in Aeronautic Clubs. Write The Goodyear Tire & Rubber Company in Akron or Los Angeles.

**GOOD YEAR**  
**AIR SHIPS**

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## Fahrig Anti-Friction Metal

*The Best Bearing Metal on the Market  
A Necessity for Aeroplane Service*



Fahrig Metal Quality has become a standard for reliability. We specialize in this one tin-copper alloy which has superior anti-friction qualities and great durability and is always uniform.

When you see a speed or distance record broken by Aeroplane, Racing Automobile, Truck or Tractor Motor, you will find that Fahrig Metal Bearings were in that motor.

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AIRPLANE ENGINES**

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West Berkeley, California**

Too much **SPEED** is the cause  
or too little of most crashes

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